

# Tension Softening of Fibre-Reinforced Cementitious Composites

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## Abstract

*The complete tension softening behaviour of short-fibre-reinforced cementitious composites which exhibit extensive matrix cracking is established by combining a fracture mechanical approach for bridged, discontinuous cracks with a statistical approach for a bridged, through-crack via consistent relationships for the bridging stress. These relationships account for adhesive and frictional bonding of fibres along a crack and are capable of accommodating the effects of fibre content and pre-peak cracking upon the fibre-binder interfacial properties. The combined approach is adapted to mortars based on a conventional high-strength cement binder and a dense, homogeneous cement-silica binder (DSP composite), and good agreement is obtained between the uniaxial tensile strength so calculated and that based on the rule of mixtures. Although the tension softening model needs many geometrical and mechanical properties of the mix constituents, as well as of the hydration products (e.g. interfaces) and the initial post-peak crack configuration, it provides a powerful tool for microstructural design of fibre-reinforced cementitious composites, particularly DSP mortars. © 1997 Elsevier Science Ltd.*

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## INTRODUCTION

Quasi-brittle materials are characterized by the localization of deformation after the attainment of tensile strength, i.e. by the continued opening of cracks and stretching of unbroken material ligaments in a narrow localization zone (band). The overall post-peak stress–inelastic displacement or tension softening response ( $w, \sigma$ ) of this zone governs many important structural aspects, such as bearing capacity and brittleness, and has accordingly received considerable attention in a number of studies. Within the framework of fracture mechanics, the tension softening response has been established by modelling the localization zone as a series of unbridged<sup>1–5</sup> (i.e. traction-free) or bridged cracks.<sup>6</sup>

In unreinforced cementitious composites, the partial bridging provided by unbroken material ligaments (typically aggregates) in the fracture process zone near the crack tips may be accounted for by considering Dugdale-type cracks<sup>1,7,8</sup> or a bridged effective through crack.<sup>9</sup> Alternatively, an appropriate value can be assigned to the fracture toughness of the material.<sup>4,10</sup> Moreover, after coalescence of the cracks, the pull-out of large aggregates along the through crack can be considered, as suggested by Huang & Li.<sup>4</sup>

For discontinuous fibre-reinforced cementitious composites, Karihaloo *et al.*<sup>6</sup> assumed the fragmented cracks to be bridged along their entire length, the continuous closing pressure being obtained by smearing the discrete closing forces provided by the fibres. However, the tension softening response was only studied prior

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to coalescence of the cracks, although the fibres usually continue to bridge after the formation of a through crack<sup>11</sup> (Fig. 1). The multiple-crack model of Karihaloo *et al.*<sup>6</sup> and the through-crack model of Li *et al.*<sup>11</sup> are appropriately combined in the following to yield the complete tension softening behaviour of fibre-reinforced cementitious composites. The cracks in both models are bridged, the bridging being provided entirely by fibres. Thus, the cracks are assumed to traverse exclusively through the matrix and not through the aggregate particles or along aggregate–matrix interfaces. This type of crack configuration is characteristic of mortars using cement, silica, and strong aggregates.

In the present study, particular emphasis is placed on mortars based on the DSP (densified systems containing homogeneously arranged ultra-fine particles) concept,<sup>12</sup> which, contrary to conventional mortars, are highly homogeneous, thereby enabling a high degree of packing. The homogeneity is achieved by effectively eliminating (reducing) the surface forces responsible for the flocculation of the particles in the fresh binder primarily through the use of surface-active agents. The high strength of the binder thus produced reduces the resistance to microcracking induced during hydration and increases the brittleness of the composite using this binder as matrix. This may be remedied through various particle (aggregate and fibre) modifications. The densely packed cement particles, partly dissolved in sufficient water to ensure water saturation, are responsible for part of the strength gain due to a high density of

chemical and physical bonds and due to the toughening (crack deflection and bridging) generated by the unhydrated parts of the cement particles. In addition, the ultrafine microsilica particles in the voids between the cement particles reduce the amount of water required for water saturation, thereby further enhancing the strength as the porosity and pore size distribution are improved.<sup>13</sup> Moreover, the density and strength of bonds increase (partly due to the replacement of weak calcium hydroxide by strong calcium silicate hydrate through the pozzolanic reaction of microsilica) and the undissolved microsilica provides additional toughening. The reduction in water demand (and the associated reduction in the amount of cement paste), together with the restraining action of the stiff, partly undissolved microsilica, reduces the overall inherent (autogeneous) shrinkage of the cement–silica binder. At the same time, however, the fine and dense microstructure produces large contractive forces<sup>12</sup> such that the shrinkage action approaches (or exceeds<sup>14</sup>) that of ordinary cement paste. Nonetheless, because of the substantially improved strength, the resistance to microcracking is lowered, i.e. the brittleness number<sup>15</sup>  $B = [g(f'_{t,matrix})^2]/(E_{matrix} G_{F,matrix})$  is increased. Here,  $g$  is the maximum size (diameter) of the aggregate particles in a matrix with uniaxial tensile strength  $f'_{t,matrix}$ , and  $E_{matrix}$  and  $G_{F,matrix}$  are Young's modulus and specific fracture energy of matrix. For structural purposes it is essential to reduce the content of shrinking binder by incorporating a high volume fraction

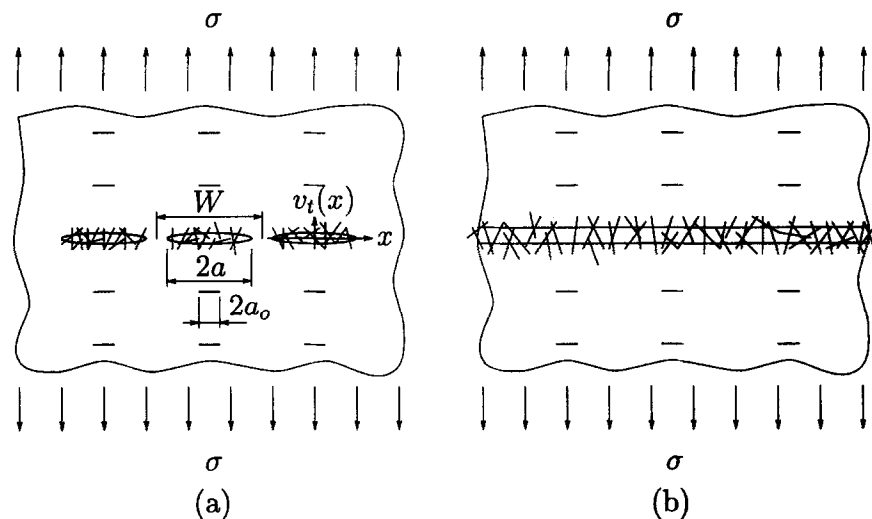


Fig. 1. (a) Crack configuration prior to coalescence of cracks, and (b) after coalescence, during tension softening of fibre-reinforced quasi-brittle materials.

of small aggregate particles to avoid a deterioration in  $B$ . Whereas this only marginally improves  $B$ , substantial improvement is achieved by incorporating fibres which primarily increase  $G_{F,matrix}$ .

Aggregate–matrix interfacial cracks (often leading to a coherent crack pattern) can be avoided, or at least reduced, by using rough aggregate particles, such as calcined bauxite.<sup>16</sup>

## TENSION SOFTENING MODEL

The model is based on effective cracks approximating the mechanical behaviour of the physical ‘crack’ (Fig. 2(a)). The physical crack is initially a collection of small unbridged cracks (including Cook–Gordon interfacial cracks) which eventually connect and perhaps propagate along the matrix–fibre interface (herein referred to as peeling — not to be confused with slip) depending on its brittleness. Crack opening due to rupture (which may be avoided if the fibres are sufficiently short) and peeling of the fibres will be assumed to be negligible, such that inelastic opening is produced only when the fibres lose their adhesive bond to the matrix and slip under establishment of frictional bonding in a zone of length  $2b$ .  $v_t(x)$  denotes the local half-opening of the effective crack.

The distribution of bridging stress along the effective crack  $p(v_t)$  is taken to conform with the slip-weakening behaviour exhibited by steel fibres,<sup>17</sup> see Fig. 2(b). For this purpose, a tri-

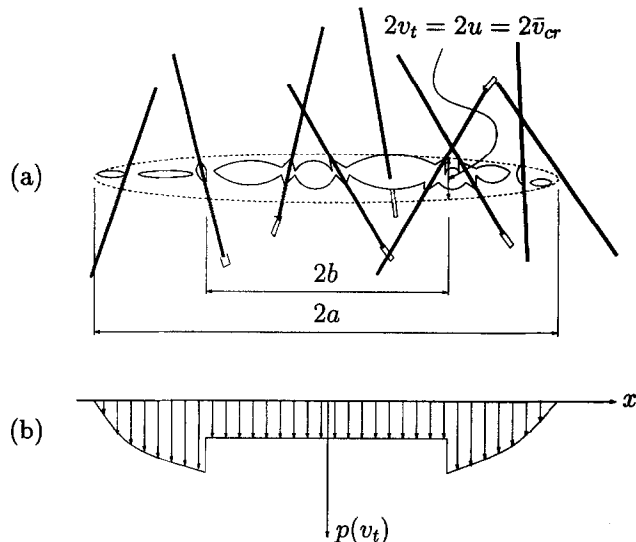


Fig. 2. (a) Effective crack (dashed line) and physical ‘crack’ in fibre-reinforced material with (b) distribution of bridging stress  $p(v_t)$  prior to coalescence of cracks.

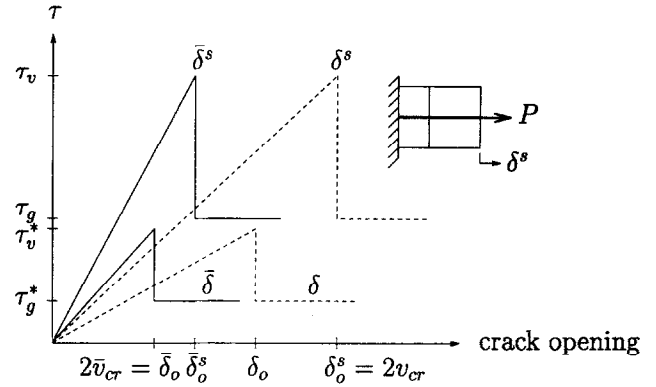


Fig. 3. Idealized interfacial matrix–fibre shear stress ( $\tau$ ) vs. crack mouth opening (elongation of fibre) ( $\delta$ ) or effective crack opening ( $\bar{\delta}$ ) relationship for single-fibre (superscript s) and multiple-fibre pull-out.

linear idealization of the interfacial shear stress–crack opening relationship is adopted (Fig. 3). As indicated, this relationship depends on the volume fraction of fibre present in the pull-out test specimen<sup>17,18</sup> and it may be expressed in terms of the mouth opening of an artificial crack (identical to the elongation of fibre)  $\delta$ , or an equivalent measure of the crack opening, e.g. that of the effective crack,  $\bar{\delta} = 2v_t$ . The latter is desirable in the present study as the multiple-crack model provides values of  $v_t$ .  $\tau_v$  ( $\tau_v^*$ ) is the adhesive bond strength,  $\tau_g$  ( $\tau_g^*$ ) the frictional bond strength and  $\delta^s$  ( $\delta$ ) the elongation of the fibre (or crack mouth opening) in a single-fibre (multiple-fibre) pull-out test. Furthermore,  $\delta_0^s$  and  $\delta_0$  are the respective crack mouth openings at initiation of fibre pull-out, whereas  $\bar{\delta}^s$ ,  $\bar{\delta}_0^s$ ,  $\bar{\delta}$  and  $\bar{\delta}_0$  denote corresponding effective crack openings.

The tri-linear relationship enables establishment of simple expressions for the bridging force provided by a fibre  $P(\bar{\delta})$  or  $P(\delta)$ . Li *et al.*<sup>11</sup> suggested a relation between  $P(\delta)$  and the bridging stress  $p(\delta)$  by considering frictionally bonded fibres along a through crack. Applying this relationship to adhesively bonded fibres and multiple effective cracks, it reads

$$p(\bar{\delta}) = \frac{V_f}{A_f} \int_{\phi=0}^{\pi/2} \int_{z=0}^{(L/2 - \bar{\delta})\cos\phi} P(\bar{\delta}, \phi, z) p(\phi) p(z) dz d\phi \quad (1)$$

where  $V_f$  is the volume fraction of fibre,  $A_f$  the cross-sectional area of a fibre,  $L$  the initial length of a fibre,  $\phi$  the orientation angle of the fibre to a plane perpendicular to the crack and  $z$  the distance from the centroid of the fibre to

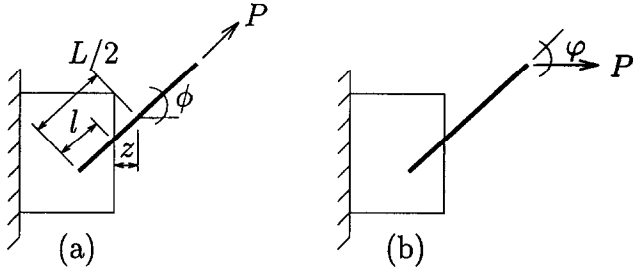


Fig. 4. (a) Single-fibre pull-out without snubbing ( $\phi = 0$ ) and (b) with snubbing ( $\phi \neq 0$ ,  $\phi = \phi$ ).

the crack plane, see Fig. 4. The probability densities are  $p(\phi) = \sin \phi$  and  $p(z) = 2/L$ . Moreover, to account for the snubbing effect (fibre dowel action) arising when a fibre is not loaded in the direction of the initial straight fibre (Fig. 4), the bridging force is increased according to  $P(\delta, \phi = \phi, l) = P(\delta, \phi = 0, l) e^{f\phi}$ , where  $\phi$  is the orientation angle of the fibre to the loading direction and  $f$  the snubbing friction coefficient.

Motivated by similar models, the bridging stress for multiple-fibre bridging will be expressed in terms of single-fibre properties  $\tau_v$  and  $\tau_g$ . This is achieved by introducing the ratio

$$\zeta = \frac{\bar{\delta}_0}{\delta_0^s} = \frac{\bar{v}_{cr}}{v_{cr}} \quad (2)$$

where  $\bar{v}_{cr}$  follows from the multiple-crack model, since it assumes slippage to be initiated immediately after reaching the ultimate tensile load, whereby the corresponding maximum half-opening of the crack equals  $\bar{v}_{cr}$ . (This type of slippage behaviour is in accord with the work of Tjiptobroto & Hansen<sup>19</sup> for DSP composites.)  $v_{cr}$  may be estimated by considering the onset of slip of a single fibre (with diameter  $d$ ) bridging perpendicularly an initially closed crack that separates the material parts shown in Fig. 5(b). A similar pull-out test set-up has been

employed by e.g. Bantia & Yan.<sup>20</sup> The estimate will be based on a stress (strength) approach; more refined fracture mechanics (energy-based) approaches are also available.<sup>18,21,22</sup> (The so-called mismatch-strain approach is also available.<sup>23</sup>) Near the crack mouth, the opening will attain a constant value,  $\delta_0^s$ , equal to the displacement of the load point. As the fibre undergoes the same amount of deformation, it will experience an average strain of  $\delta_0^s/L$  and a strain of  $2\delta_0^s/L$  across the crack provided peeling is limited. The bridging stress is  $P_0/(\pi d^2/4)$  with  $P_0 = \tau_g \pi d l' = \tau_g \pi d l$  — the bridging force carried by the fibre at the attainment of slip over the entire embedded length  $l'$ . Owing to peeling, the short embedded fibre segment is reduced from  $l$  to  $l'$ . For a linear-elastic fibre, therefore, the critical half-crack opening at onset of full slippage becomes  $v_{cr} = 1/2 \delta_0^s = (\tau_g L l)/(E_f d)$ . Similar expressions have been reported by Li,<sup>24</sup> Li *et al.*,<sup>25</sup> and Balaguru & Shah.<sup>18</sup> The interfacial shear stress prior to slippage is given by  $\tau = \delta \tau_v^*/(\delta_0^s \zeta)$ , resulting in  $P = (\tau_v^*/\zeta)[E_f d^2 \pi/(2\tau_g L)] e^{f\phi} \delta$ , while during slippage  $P = \tau_g \pi d l [1 - (\delta - \delta_0)/l] e^{f\phi}$ . In performing the integrations indicated in eqn 1, it is assumed that  $\zeta$  attains its mathematical expectation  $E(\zeta) = \bar{v}_{cr}/E(v_{cr})$ , where

$$\begin{aligned} E(v_{cr}) &= \frac{\tau_g L}{E_f d} E(l) = \frac{\tau_g L}{E_f d} \left[ \frac{L}{2} - \frac{E(z)}{\cos \phi} \right] \\ &= \frac{1}{4} \frac{\tau_g L^2}{E_f d} \end{aligned} \quad (3)$$

following the observation by Li *et al.*<sup>11</sup> that  $l = L/2 - z/\cos \phi$  has a distribution according to

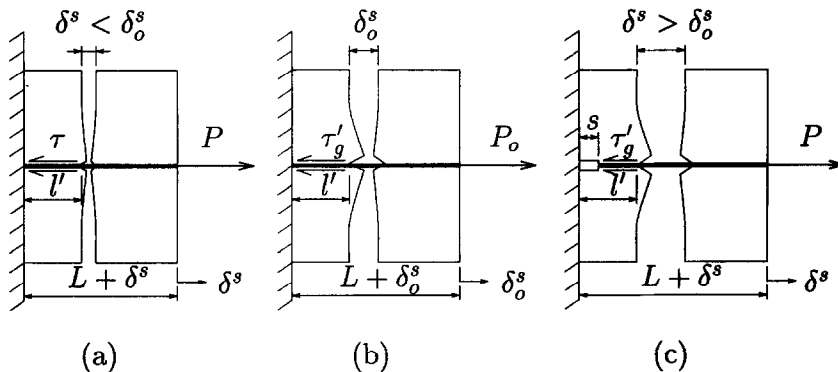


Fig. 5. Single-fibre pull-out (a) prior to slip, (b) at onset of slip, and (c) during slip.

the probability densities of  $\phi$  and  $z$ . Further, making use of the observation<sup>17</sup>  $\tau_v^*/\bar{v}_{cr} \approx \tau_v/v_{cr}$

and the assumption  $\tau_g^*/\bar{v}_{cr} \approx \tau_g/v_{cr}$ , we eventually have (see Appendix A for details)

$$p(v_t) = \begin{cases} 2V_f \frac{\tau_v}{\tau_g} E_f h \left( \frac{1}{L} v_t - \frac{4}{L^2} v_t^2 \right) \approx 2V_f \frac{\tau_v}{\tau_g} E_f \frac{h}{L} v_t, & v_t < \bar{v}_{cr} \\ V_f \tau_g \frac{L}{d} \frac{1}{2} h E(\zeta) \left[ -2 \left( \frac{2v_t}{L/2} \right) + (2-\alpha) \left( \frac{2v_t}{L/2} \right)^2 + \alpha \right] \approx V_f \tau_g \frac{L}{d} \frac{1}{2} h \zeta, & v_t \geq \bar{v}_{cr} \end{cases} \quad (4)$$

Here, the snubbing factor  $h = 2/(4+f^2)(e^{\pi/2}+1)$  and  $\alpha = 1 + \{[2\tau_g LE(\zeta)]/(E_f d)\} (\approx 1)$ . The exact  $p(v_t)$  expression for  $v_t \geq \bar{v}_{cr}$  is identical to Li's<sup>24</sup> tension softening relation  $p(\delta)$  when  $\alpha = 1$  and  $\bar{\delta} = 2v_t \approx \delta$  for large crack openings. It should be clear, however, that  $\bar{\delta}$  is not the sought inelastic crack opening  $w$  but rather the elastic ( $\bar{\delta}_0$ )+inelastic crack opening ( $w$ ) such that the tension softening relation after the through crack has formed at  $w = w_k$  becomes

$$\sigma(w) = \sigma(\bar{\delta} - \bar{\delta}_0) = V_f \tau_g \zeta \frac{L}{d} \frac{1}{2} \times \left[ -\frac{4}{L} w + \frac{4}{L^2} (2-\alpha)w^2 + \alpha \right], \quad w > w_k \quad (5)$$

with  $\sigma(w) = 0$  for  $w = (L/2)[1 - \sqrt{1 - \alpha(2-\alpha)}]$  ( $= L/2$ , the largest fibre embedment length, for  $\alpha = 1$ ).

We finally turn our attention to the tension softening response prior to coalescence of the cracks, i.e.  $w \leq w_k$ . For a given crack configuration ( $a, b, W$ ) and applied stress  $\sigma$  the model by Karihaloo *et al.*<sup>6</sup> provides the corresponding stress intensity factor  $K_I$  and crack openings  $v_t$ . In particular, at peak load ( $\sigma = f'_b$ ,  $a = a_0$ ,  $b = 0$ ), we determine the fracture toughness  $K_{Ic} = K_I$  and  $\bar{v}_{cr}$ . For  $K_I = K_{Ic}$  in the entire post-peak regime, the model predicts a decreasing  $w$  when the crack propagates beyond a certain crack length. In reality, however,  $w$  increases monotonically. For an ideal brittle material this is believed to be accomplished by the deviation of the cracks from their plane<sup>26</sup>, whereby  $K_I$  decreases below the value predicted by the original model, whereas for quasi-brittle materials, such as those studied herein, the

resistance to crack propagation  $K_{Ic}$  is enhanced presumably by the interaction of the process zones. The increase in  $K_{Ic}$  is manifested through a change in sign of the curvature  $\partial^2 \sigma / \partial w^2$  of ( $\sigma, w$ ) diagram at  $a > a_i$  (corresponding to points of instability). Thus, for  $a > a_i$ ,  $K_{Ic}$  is increased ( $K_{Ic} = 0.01 \text{ MPa m}^{0.5}$ ) such that instability is avoided, i.e.  $\partial^2 \sigma / \partial w^2$  maintains the same sign throughout tension softening. It is found, however, that this condition can no longer be satisfied when the cracks are very close to one another, i.e. when  $a > a_c$ . At  $a = a_c$ , only a few fibres along the discontinuous crack segments remain elastically bonded to the matrix. Hence, an instantaneous drop in the stress is to be anticipated (similar to the pull-out of single fibres, see Fig. 3) when the crack fragments link up to form a through crack. Ortiz<sup>2</sup>, amongst others, claims that a sudden drop is inevitable even without fibres when the discontinuous crack fragments link up to form a through crack.

The tension softening relation, i.e. the inelastic deformation of the localization zone, is given by

$$w(\sigma) = \frac{2}{W} \left[ \int_{-a}^a v_t(\sigma, x) dx - \int_{-a_0}^{a_0} v_t(f'_b, x) dx \right] \quad (6)$$

as the opening of the cracks prior to the attainment of peak load is assumed to be purely elastic and the additional opening after the attainment of peak load is assumed to be purely inelastic. For a given half-length of crack  $a$ ,  $w$  corresponds to the values of the unknown stress and of the sliding zone  $b$  at which the following two conditions for crack growth are simultaneously satisfied

$$K_I(\sigma, a; b) = K_{Ic}(a_0, W, \tau_v, \tau_g, V_f, f, L, d, E_f, E, f'_t) \quad (7)$$

$$u(\sigma, a; b) = \bar{v}_{cr}(a_0, W, \tau_v, \tau_g, V_f, f, L, d, E_f, E, f'_t) \quad (8)$$

Here  $u(\sigma, a; b)$  is the total crack opening at the ends of the assumed sliding zone.

Nielsen<sup>27</sup> has shown that the modulus of elasticity of a two-phase isotropic composite is given by

$$E = E_m \frac{n + \Theta + V\Theta(n-1)}{n + \Theta - V(n-1)} \quad (9)$$

where  $V$  is the volume fraction of the discrete phase,  $\Theta$  is a geometry function accounting for the configuration of the discrete phase and  $n$  is the ratio of the modulus of elasticity of the discrete phase to the modulus of elasticity of the continuous phase  $E_m$ . Cement paste, mortar and concrete can be regarded as two-phase composites with the modulus of elasticity given by<sup>28</sup> eqn 9. For cement paste ( $E = E_p$ ), a distinction is made between water/cement ratios below and above 1.2  $\rho_w/\rho_c$  ( $\rho_w$  is the density of water;  $\rho_c$  is the density of unhydrated cement particles). In the former range, the cement paste is considered as a collection of unhydrated cement particles with modulus of elasticity  $E_u$  embedded in partially hydrated cement gel with  $E_m = 27200h$  MPa, where  $h$  is taken as the relative degree of hydration. The volume fraction of unhydrated cement particles is

$$V = \frac{1 - 0.83h(w/c)(\rho_c/\rho_w)}{1 + (w/c)(\rho_c/\rho_w)} \quad (10)$$

and

$$\Theta = 0.5[\eta_u\sqrt{1-V(1-n)} + \sqrt{\eta_u^2(1-V)(1-n)^2 + 4n}] \quad (11)$$

where  $\eta_u$  is a shape factor of an unhydrated cement particle. For  $w/c > 1.2\rho_w/\rho_c$ , the cement paste is regarded as a collection of capillary pores embedded in solid, hydrated cement gel. Now,  $n = 0$ ,  $\Theta = \eta_k(1-V)$  with  $\eta_k$  being a shape factor of a capillary pore and the volume fraction of capillary pores

$$V = \frac{w/c - 1.2h\rho_w/\rho_c}{w/c + \rho_w/\rho_c} \quad (12)$$

For mortar ( $E = E_r$ ),  $E_m = E_p$ ,  $n$  is the ratio of modulus of elasticity of the fine aggregate particles  $E_s$  to  $E_p$  and  $V = V_s$ . Moreover,<sup>29</sup>

$$\Theta = \frac{1}{2} \{q + \sqrt{q^2 + 4n[1 - \eta_s(1 - V_s) - \eta_s(V_s - 1)]}\} \quad (13)$$

Here,  $q = \eta_s(1 - V_s) + n\eta_s(V_s - 1)$  and

$$\eta_s = \frac{3A_s(1+A_s)}{1+A_s+4A_s^2} \quad (14)$$

with  $A_s$  being the aspect ratio of the fine aggregate particles. Having determined the modulus of elasticity of the matrix phase in a fibre-reinforced composite from the expressions in eqn 9–14, eqn 9 can again be used to find  $E$  of the fibre composite. Now, let  $V_f = V$  and<sup>30</sup>

$$\Theta = \frac{1}{2} [\mu + n\mu' + \sqrt{(\mu + n\mu')^2 + 4n(1 - \mu - \mu')}] \quad (15)$$

in which

$$\mu = \mu_0 \frac{c_d - V_f}{c_d}$$

$$\mu' = \min \left( \mu_0 \frac{V_f - c_D}{c_d}, 1 \right)$$

$$\mu_0 = \frac{3A(1+A)}{1+A+4A^2}$$

$$c_d = \frac{\mu_0}{1 + \mu_0}$$

$$c_D = \frac{(2c_d)^{10}}{2} \quad (16)$$

and the aspect ratio  $A = L/d$  and  $n = E_f/E_b$ .

The uniaxial tensile strength of the fibre composite follows from

$$f'_t = \frac{P_f + P_b}{W^2} \quad (17)$$

where the contributions from the fibres and the remaining material are respectively

$$P_f = p(v_t = w_t/2)W^2 = V_f \frac{\tau_v}{\tau_g} E_f \frac{h}{L} w_t(\sigma = f'_t)W^2$$

(18)

$$P_b = f'_{t,b}(1 - V_f)(W^2 - 4a_0^2) \quad (19)$$

The total effective deformation of the localization zone is

$$w_t(\sigma) = \frac{w_{lig}(W - 2a)}{W} + \frac{2}{W} \int_{-a}^a v_t(\sigma, x) dx \quad (20)$$

In the unbroken ligaments, strain compatibility between the matrix and the fibres is assumed, such that  $w_{lig} = f'_{t,b}L/E_b$ . The uniaxial tensile strength for a particulate composite as established by Huang & Li<sup>4</sup> yields

$$f'_{t,b} = \frac{K_{Ic,p}}{\sqrt{\pi g/2}} \eta$$

where  $\eta$  is the toughening ratio due to various toughening mechanisms,  $K_{Ic,p}$  is the fracture toughness of the binder and  $g$  is the maximum aggregate size (diameter). For a material with a strong aggregate-matrix interface and small, strong aggregate particles with tensile strength  $f'_{t,a}$  and average size (diameter)  $g_{av}$  the toughening may be assumed to be caused by crack deflection, bridging and trapping, whereby<sup>4,31</sup>

$$\eta = \sqrt{1.0 + 0.87V} \times$$

$$\sqrt{\chi^2 + \frac{E'_b(\pi/2)f_{t,a}^2g_{av}V(1 - \sqrt{V})(1 - V)(1 - v_p^2)}{E_b(1 - v^2)(K_{Ic,p})^2}} \quad (22)$$

where

$$\chi = \left(1 - \frac{(1 - V)\pi/4}{\ln\{[1 + \cos(\pi V/2)]/\sin(\pi V/2)\}}\right)^{-1}$$

$$\frac{E'_b}{E_b} = 1 - \frac{\pi^2}{16}(1 - v^2)V$$

$E'_b$  is the modulus of elasticity of the composite exclusive of the fibres with interfacial cracks, whereas  $E_b$  is that without interfacial cracks.  $V$  is the volume fraction of aggregates. For the cement binder, Lange-Kornbak & Karihaloo<sup>32</sup>

**Table 1.** Values of microstructural parameters for selected mortars

Microstructural parameter	Mix I	Mix II
$V_c$	0.4741	0.4741
$V_{ms}$	0.1627	0
$V_f$	variable	variable
$g$ (mm)	4.0	4.0
$g_{av}$ (mm)	1.0	1.0
$a_0$ (mm)	variable	variable
$W$ (mm)	50.0	50.0
$f'_{t,a}$ (MPa)	10.0	10.0
$E_a$ (MPa)	65000	65000
$E_f$ (GPa) <sup>33</sup>	210.0	210.0
$L$ (mm)	20	20
$d$ (mm)	0.4	0.4
$\tau_v$ (MPa) <sup>12,18,21</sup>	9.0	3.0
$\tau_p$ (MPa) <sup>12,18,21</sup>	5.0	2.5
$f'_{f1,24}$	0.9	0.75

proposed the empirical relation between  $K_{Ic,p}$  and water to cement (w/c) ratio

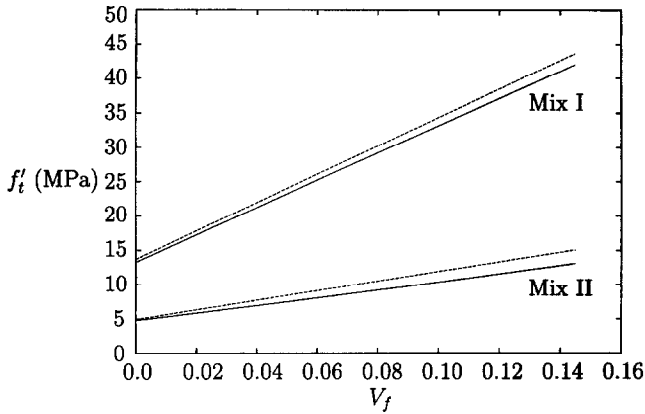
$$K_{Ic,p} = 0.6125 - 0.85w/c \text{ MPa m}^{0.5} \quad (23)$$

for  $0.2 \leq w/c \leq 0.45$ .

## DISCUSSION

In the following, two mixes with the microstructural parameters given in Table 1 are compared. Mix I is a cement-silica-based DSP binder, whereas mix II is based on a conventional high-strength cement binder. Both mixes contain round, plain straight steel fibres. The composition of the binder follows from  $1 = V_{ms} + V_c + V_w$  with  $V_{ms}$  the volume fraction (packing density) of microsilica particles,  $V_c$  that of cement particles and  $V_w$  that of water (and dispersing agent). The volume fraction of fine aggregate is given by  $(1 - V_p - V_f)/(1 - V_f)$  where the volume fraction of paste is  $V_p = 0.5$  for all  $V_f$  used in the tests by Tjiptobroto & Hansen.<sup>19</sup> Note that  $w/c = V_w\rho_w/(V_c\rho_c)$ . It is assumed that  $K_{Ic,p} = 0.5 \text{ MPa m}^{0.5}$  and  $E_p = 35000 \text{ MPa}$  for mix I and that crack trapping is absent (i.e.  $\chi = 1.0$ ) in mix II. The study of Tjiptobroto & Hansen<sup>19</sup> suggests that the crack spacing  $W$  is only marginally affected by the variation in paste composition studied here. Following the work of Karihaloo *et al.*<sup>6</sup>  $W = 50 \text{ mm}$  is chosen for both mixes. Moreover, any alteration in  $W$  due to a variation in  $V_f$  has been neglected.

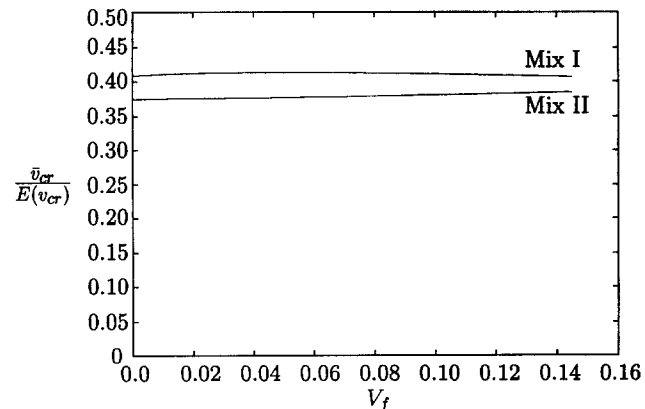
A comparison of eqn 17 with the rule of mixtures<sup>18,21</sup>



**Fig. 6.** Uniaxial tensile strength as a function of volume fraction of fibre according to eqn 17 (solid line) and eqn 24 (dashed line) ( $a_0 = 4.5$  mm).

$$f'_t = \gamma f'_{t,b}(1 - V_f) + \beta \tau_v V_f \frac{L}{d} \quad (24)$$

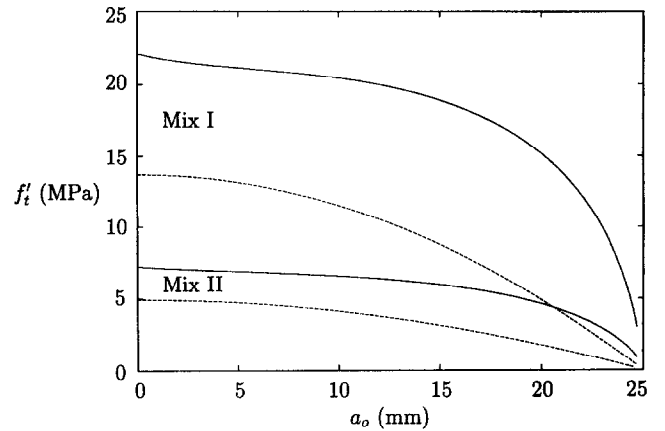
appears to lend considerable support to the proposed model, as good agreement is obtained for both mixes in the entire range of  $V_f$  values, see Fig. 6. In the comparison  $\gamma = 1$ , so that  $f'_t = f'_{t,b}$  at  $V_f = 0$ , and  $\beta = 0.5$  are in agreement with the use of randomly distributed short fibres.<sup>6</sup> The rise in  $f'_t$  with an increase in  $V_f$  is predominantly an effect of  $V_f$  as  $f'_{t,b}$  and  $\bar{v}_{cr}$  (see Fig. 7) are found to be only marginally influenced by this parameter;  $f'_{t,b}$  varies from 4.95 MPa (13.69 MPa) at  $V_f = 0$  to 4.87 MPa (12.85 MPa) at  $V_f = 0.145$  for Mix I (II). For practical conventional mixes, the homogeneity of the paste and the random distribution of fibres cannot be maintained at more than a few percent fibres so that the increase in  $f'_t$  is less pronounced than shown in Fig. 6; even a decrease in  $f'_t$  may be experienced. In other words, at increasing  $V_f$  the numerical coeffi-



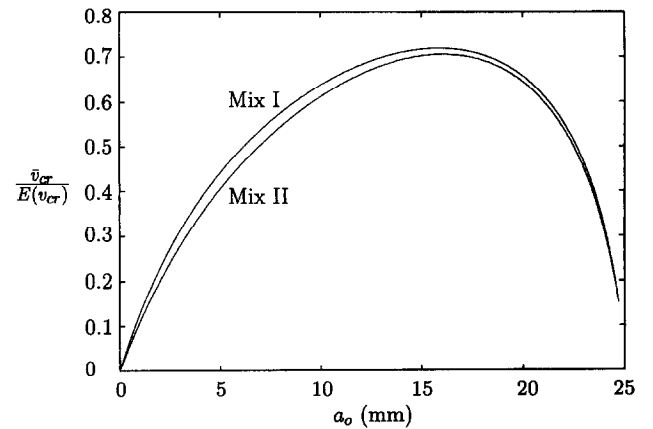
**Fig. 7.** Ratio of  $\bar{v}_{cr}$  to expectation of  $v_{cr}$  as a function of volume fraction of fibre ( $a_0 = 4.5$  mm).

cients  $\gamma$  and  $\beta$  in eqn 24 are reduced. In the proposed theoretical model, however,  $K_{Ic,p}$ ,  $\tau_v$  and  $\tau_g$  would decrease with increasing  $V_f$  and the probability densities  $p(\phi)$  and  $p(z)$  in eqn 1 would change according to the actual statistical distribution of the fibres. For DSP mixes, on the other hand, measurements of the flexural tensile strength, compressive strength and specific fracture energy indicate that the homogeneity of the paste and the random distribution of fibres is maintained up to at least<sup>19,34</sup>  $V_f = 0.12$ .

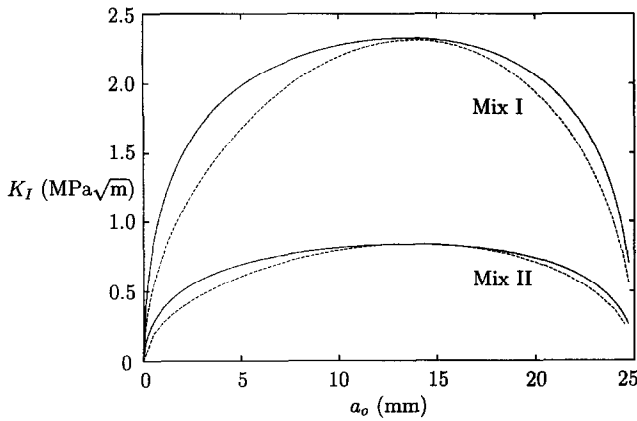
The loss in strength resulting from increased crack length  $a_0$  shown in Fig. 8 can be attributed to the reduction in  $P_b$ , which is only partly compensated by an increase in  $P_f$  ( $\bar{v}_{cr}$ ) at small  $a_0$  (Fig. 9). The variation of  $\bar{v}_{cr}$  with  $a_0$  noticeable in Fig. 9 results from the averaging implied in eqn 20.



**Fig. 8.** Variation of uniaxial tensile strength with half-length of crack  $a_0$  at ultimate tensile load for bridged ( $V_f = 0.04$ , solid line) and unbridged cracks ( $V_f = 0$ , dashed line).



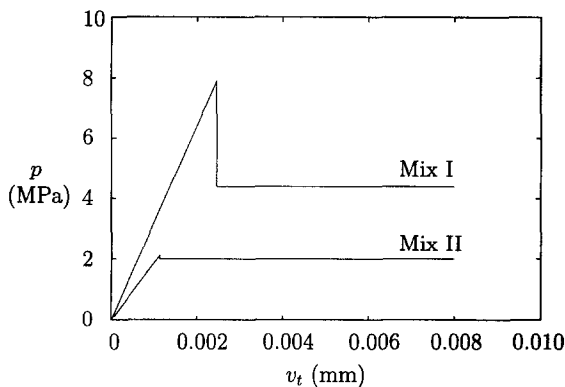
**Fig. 9.** Ratio of  $\bar{v}_{cr}$  to expectation of  $v_{cr}$  as a function of half-length of crack  $a_0$  at ultimate tensile load ( $V_f = 0.04$ ).



**Fig. 10.** Variation of stress intensity factor  $K_I$  with half-length of crack  $a_0$  at ultimate tensile load for bridged ( $V_f = 0.04$ , solid line) and unbridged cracks ( $V_f = 0$ , dashed line).

Further support to the model is provided in Fig. 10, which shows that the variation of  $K_I$  in the post-peak regime (where  $K_I = K_{Ic}$ ) with  $a_0$  follows the trend established for unbridged cracks. In the latter case,<sup>35</sup>  $K_I = K_{Ic} = \sqrt{W \tan(\pi a_0/W) f'_t}$ . As expected,  $K_I$  ( $K_{Ic}$ ) is the greatest for the densest mix and the largest  $V_f$ . At large  $a_0$ , the reduction in  $K_{Ic}$  caused by the lower  $f'_t$  and bridging stress (see Fig. 9) overshadows the enhancement arising from increases in  $a_0$ .

Figure 11 shows the bridging stress–total crack opening relationship according to eqn 4. For other values of the half-length of crack  $a_0$  at peak load, the modified relationship follows from the value of  $E(\zeta)$  as  $\bar{v}_{cr}$ , unlike  $v_{cr}$ , is affected by  $a_0$ . With  $\bar{v}_{cr} = E(v_{cr})E(\zeta)$ , the maximum and the residual bridging stress follow from eqn 4. Figure 9 shows that the bridging stresses increase with increasing  $a_0$  up to about 16 mm, so that the fibres provide the most efficient bridging as expressed by  $E(\zeta)$ , when

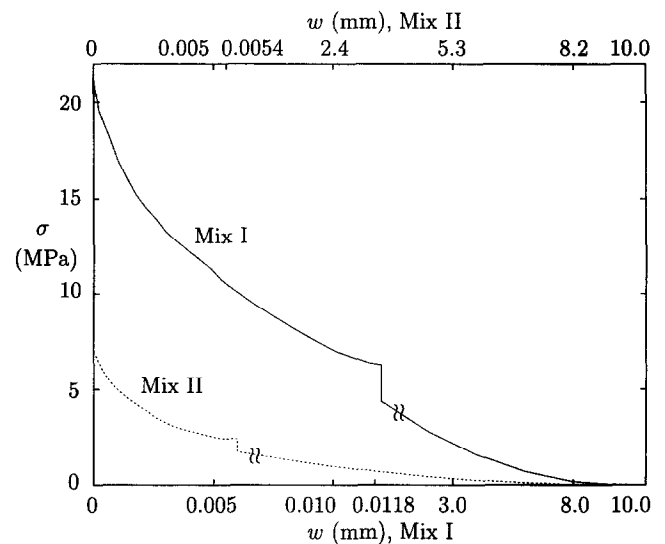


**Fig. 11.** Bridging stress–total crack opening relationship ( $a_0 = 4.5$  mm).

medium-sized cracks are present at ultimate load. The same figure provides the crack opening  $\bar{v}_{cr}$  required for crack propagation at a given crack length. It also reflects the assumption that fibre pull-out is initiated at the onset of tension softening where the maximum crack opening is  $\bar{v}_{cr}$  and not on the descending branch when the maximum crack opening reaches  $v_{cr}$ .  $a_0$  is controlled by the brittleness of the binder and  $V_f$ ; a higher brittleness and lower content of fibre result in larger crack lengths ( $a_0$ ). However, Fig. 9 does not take into account that  $\tau_g(E(v_{cr}))$  is affected by the brittleness (composition) of the binder associated with variations in  $a_0$ .

The tension softening responses are shown in Fig. 12. For Mix I (II),  $K_{Ic}$  has increased from  $1.93 \text{ MPa m}^{0.5}$  ( $0.67 \text{ MPa m}^{0.5}$ ) at  $a_i = 0.0185$  mm ( $0.0195$  mm) to  $6.09 \text{ MPa m}^{0.5}$  ( $2.21 \text{ MPa m}^{0.5}$ ) at  $a_c = 0.02493$  mm. It is noted that the fracture energy  $G_F$  of the fibre-reinforced DSP material is  $16000 \text{ Nm/m}^2$ , which is within the range  $5000\text{--}40000 \text{ Nm/m}^2$  reported for these materials.<sup>34</sup>

Two aspects of the above calculations need to be stressed. First, there are 32 integration points per half-crack length. As the bridging zone is restricted to grow discontinuously from one integration point to the next,  $K_I$  and  $u$  are allowed to deviate 5% from the target values ( $K_{Ic}$  and  $\bar{v}_{cr}$  respectively) in order to be able to satisfy the crack growth criteria in eqns 7–8. Secondly, a unique tension-softening curve is determined by letting the stress to decrease, i.e. letting  $a$  increase from  $a_0$ , as the snap-back (instability) in the tail region predicted by the



**Fig. 12.** Tension softening response.

original model represents a series of bifurcation points that can lead to different tension-softening curves when the stress is increased, i.e.  $a$  is decreased towards  $a_0$ .

## CONCLUSIONS

The present study has established the complete tension softening response of short-fibre-reinforced cementitious composites in which the cracks in the localization zone are bridged along their entire length by adhesively or frictionally bonded fibres. This is achieved by using a consistent bridging-stress relationship in two existing models. The first model, suggested by Karihaloo *et al.*,<sup>6</sup> is applicable to the initial growth of discontinuous crack segments, whereas the other model by Li *et al.*<sup>11</sup> describes the tension softening behaviour after coalescence of the cracks into a through crack. The former model is additionally modified by imposing a stability criterion on crack growth which involves an increase in fracture toughness immediately before the cracks link up to form a through crack.

Besides giving the specific fracture energy  $G_F$  of the composite, the proposed model is capable of predicting the uniaxial tensile strength  $f_t$  which is in good agreement with the rule of mixtures. Nonetheless, the assumption that fibre pull-out is initiated at onset of tension softening needs verification. Further improvement of the model is possible by a more accurate prediction of the initial post-peak crack configuration.

Having established the relationships between the effective properties (i.e. direct tensile strength  $f_t$ , fracture energy  $G_F$  and modulus of elasticity  $E$ ) of a fibre-reinforced concrete mix and its microstructural parameters (e.g. fibre-matrix interfacial adhesive or frictional bond strength, length of fibres, diameter of fibres, volume fraction of fibre and volume fraction of aggregates) it is possible to design such composites for simultaneous minimum brittleness, i.e. maximum characteristic length  $EG_F/f_t^2$ , and maximum tensile strength. This can be accomplished with the use of mathematical programming techniques, as demonstrated for plain concrete.<sup>32</sup> The results for short-fibre-reinforced cementitious composites based on the DSP concept will be reported in a future communication.

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## APPENDIX A

### Bridging stress relationship outside the slip zone

The bridging force carried by a fibre with short embedment length  $l$ , oriented at an angle  $\phi$  and stressed  $\bar{\delta}$  is  $P(\bar{\delta}, \phi, l) = \tau \pi d l e^{f\phi}$ , where the interfacial shear stress  $\tau = (\tau_v^*/\bar{\delta}_0)\bar{\delta} = [\tau_v^*/(\zeta\bar{\delta}_0^s)]\bar{\delta} = \tau_v\bar{\delta}/\bar{\delta}_0^s$  with  $\bar{\delta}_0^s = 2\tau_v L l / (E_f d)$ .

From eqn 1, it follows that the bridging stress is

$$p(\bar{\delta}) = \frac{V_f}{A_f} \frac{\tau_v}{\tau_g} \frac{d^2}{L^2} E_f \pi \bar{\delta} \int_{\phi=0}^{\pi/2} \int_{z=0}^{(L/2-\bar{\delta})\cos\phi} e^{f\phi} \sin\phi \, dz \, d\phi \quad (25)$$

where

$$\int_{z=0}^{(L/2-\bar{\delta})\cos\phi} e^{f\phi} \sin\phi \, dz = e^{f\phi} \sin\phi [z]_{z=0}^{(L/2-\bar{\delta})\cos\phi} = \frac{1}{2} \left( \frac{1}{2} L - \bar{\delta} \right) e^{f\phi} \sin 2\phi \quad (26)$$

such that

$$\int_{\phi=0}^{\pi/2} \int_{z=0}^{(L/2-\bar{\delta})\cos\phi} e^{f\phi} \sin\phi \, dz \, d\phi = \frac{1}{2} \left( \frac{1}{2} L - \bar{\delta} \right) \int_{\phi=0}^{\pi/2} e^{f\phi} \sin 2\phi \, d\phi \quad (27)$$

with

$$\int_{\phi=0}^{\pi/2} e^{f\phi} \sin 2\phi \, d\phi = \frac{1}{2} \left( \frac{1}{2} L - \bar{\delta} \right) \left[ \frac{e^{f\phi} (f \sin 2\phi - 2 \cos 2\phi)}{4 + f^2} \right]_{\phi=0}^{\pi/2} = \frac{2}{4 + f^2} (e^{f\pi/2} + 1) = h \quad (28)$$

where  $h$  is the snubbing factor.

Finally,

$$p(\bar{\delta}) = \frac{V_f}{A_f} \frac{\tau_v}{\tau_g} \frac{d^2}{L^2} E_f \pi \bar{\delta} \frac{1}{2} \left( \frac{L}{2} - \bar{\delta} \right) h = 2V_f \frac{\tau_v}{\tau_g} E_f h \left( \frac{1}{L} v_t - \frac{4}{L^2} v_t^2 \right) \quad (29)$$

since  $\bar{\delta} = 2v_t$  and  $A_f = d^2\pi/4$ .

## APPENDIX B

### Bridging stress relationship inside the slip zone

In this case, the reduction of the short embedment length is equal to the slip  $\bar{\delta} - \bar{\delta}_0$ , thereby producing a bridging force  $P(\bar{\delta}, \phi, l) = \tau_g^* \pi d [1 - (\bar{\delta} - \bar{\delta}_0)] e^{f\phi}$ . With  $\tau_g^* = \tau_g \zeta$ ,  $\bar{\delta}_0 = \zeta \delta_0^s = \zeta (2\tau_g L l) / (E_f d)$  and  $l = L/2 - z/\cos \phi$ , the force is given by  $P(\bar{\delta}, \phi, l) = \tau_g \zeta \pi d (\alpha L/2 - \bar{\delta} - \alpha z/\cos \phi) e^{f\phi}$ , where  $\alpha = 1 + 2\tau_g L \zeta / E_f d$ .

Following eqn 1, the bridging stress can be written

$$p(\bar{\delta}) = \frac{V_f}{A_f} \tau_g \zeta \frac{2}{L} \pi d \int_{\phi=0}^{\pi/2} \int_{z=0}^{(L/2-\bar{\delta})\cos \phi} \left( \alpha \frac{L}{2} - \bar{\delta} - \alpha \frac{z}{\cos \phi} \right) e^{f\phi} \sin \phi \, dz \, d\phi$$

where

$$\begin{aligned} \int_{z=0}^{(L/2-\bar{\delta})\cos \phi} \left( \alpha \frac{L}{2} - \bar{\delta} - \alpha \frac{z}{\cos \phi} \right) e^{f\phi} \sin \phi \, dz &= e^{f\phi} \sin \phi \left( \alpha \frac{L}{2} z - \bar{\delta} z - \frac{1}{2} \alpha \frac{1}{\cos \phi} z^2 \right) \Big|_{z=0}^{(L/2-\bar{\delta})\cos \phi} \\ &= \left[ -\frac{1}{2} L \bar{\delta} + \frac{1}{2} (2 - \alpha) \bar{\delta}^2 + \frac{1}{8} L^2 \alpha \right] e^{f\phi} \sin \phi \cos \phi \end{aligned} \quad (31)$$

such that

$$\begin{aligned} \int_{\phi=0}^{\pi/2} \int_{z=0}^{(L/2-\bar{\delta})\cos \phi} \left( \alpha \frac{L}{2} - \bar{\delta} - \alpha \frac{z}{\cos \phi} \right) e^{f\phi} \sin \phi \, dz \, d\phi &= \left[ -\frac{1}{2} L \bar{\delta} + \frac{1}{2} (2 - \alpha) \bar{\delta}^2 \right. \\ &\quad \left. + \frac{1}{8} L^2 \alpha \right] \int_{\phi=0}^{\pi/2} e^{f\phi} \sin \phi \cos \phi \, d\phi = \left[ -\frac{1}{2} L \bar{\delta} + \frac{1}{2} (2 - \alpha) \bar{\delta}^2 + \frac{1}{8} L^2 \alpha \right] \frac{1}{2} h \end{aligned} \quad (32)$$

Finally

$$\begin{aligned} p(\bar{\delta}) &= \frac{V_f}{A_f} \tau_g \zeta \frac{2}{L} \pi d \frac{1}{2} h \left[ -\frac{1}{2} L \bar{\delta} + \frac{1}{2} (2 - \alpha) \bar{\delta}^2 + \frac{1}{8} L^2 \alpha \right] = \frac{4V_f}{d} \tau_g \zeta \frac{1}{L} h \left[ -\frac{1}{2} L \bar{\delta} + \frac{1}{2} (2 - \alpha) \bar{\delta}^2 \right. \\ &\quad \left. + \frac{1}{8} L^2 \alpha \right] = V_f \tau_g \frac{L}{d} \frac{1}{2} h \zeta \left[ -\frac{4}{L} \bar{\delta} + \frac{4}{L^2} (2 - \alpha) \bar{\delta}^2 + \alpha \right] = V_f \tau_g \frac{L}{d} \frac{1}{2} h \zeta \left[ -2 \left( \frac{2v_t}{L/2} \right) + (2 - \alpha) \left( \frac{2v_t}{L/2} \right)^2 + \alpha \right] \end{aligned} \quad (33)$$

which is identical to Li's<sup>24</sup> tension softening expression, when  $\alpha = 1$ ,  $\zeta = 1$  and  $\bar{\delta} = \delta$ .

**NOMENCLATURE**

$a$	half-length of cracks
$a_c$	half-length of cracks at the instant of crack coalescence
$a_i$	half-length of cracks at onset of instability
$a_0$	half-length of cracks at the attainment of $f'_t$
$A$	aspect ratio
$A_f$	cross-sectional area of a fibre
$b$	half-length of slip zone
$B$	brittleness number
$d$	diameter of fibre
$E$	modulus of elasticity of fibre composite
$E_f$	modulus of elasticity of fibre
$E_m$	modulus of elasticity of the continuous phase in two-phase composite
$E_p$	modulus of elasticity of the paste
$E(l), E(v_{cr})$	mathematical expectation of $l$ and $v_{cr}$ respectively
$E(z), E(\zeta)$	mathematical expectation of $z$ and $\zeta$ respectively
$f$	snubbing friction coefficient
$f'_t$	direct tensile strength
$f'_{t,a}$	direct tensile strength of aggregate particle
$f'_{t,b}$	direct tensile strength of composite without fibres
$g$	maximum aggregate size
$g_{av}$	average aggregate size
$G_F$	specific fracture energy
$h$	snubbing factor $[2/(4+f^2)](e^{f\pi/2}+1)$
$K_I$	stress intensity factor in mode I
$K_{Ic}$	critical stress intensity factor in mode I (fracture toughness)
$K_{Ic,p}$	fracture toughness of hardened binder
$l$	short embedded fibre segment
$L$	length of a fibre
$n$	ratio of modulus of elasticity of the discrete phase to that of the continuous phase
$p(z)$	probability density for $z$
$p(\phi)$	probability density for $\phi$
$P$	force carried by a fibre
$P_b$	force transmitted by the localization zone due to unbroken matrix ligaments
$P_f$	force transmitted by the localization zone due to fibres
$P_0$	force carried by a fibre at the attainment of fully developed slip

$u$	total half-opening of crack where slip is assumed to initiate
$v_{cr}$	local half-opening of crack in single-fibre pull-out at onset of slip
$\bar{v}_{cr}$	effective half-opening of crack where slip initiates
$v_t$	local half-opening of crack
$V$	volume fraction of the discrete phase in two-phase composite
$V_c$	volume fraction of cement particles
$V_f$	volume fraction of fibre
$V_{ms}$	volume fraction of microsilica particles
$V_p$	volume fraction of paste
$V_w$	volume fraction of water and dispersing agent
$w$	inelastic deformation of the localization zone
$w_k$	inelastic deformation of the localization zone at the instant of crack coalescence
$w_t$	total deformation of the localization zone
$w/c$	water/cement ratio
$W$	crack spacing in the localization zone
$z$	distance from the centroid of the fibre to the crack plane

**Greek letters**

$\alpha$	factor $1+[2\tau_g LE(\zeta)]/(E_f d)$
$\beta$	constant
$\chi$	crack trapping parameter
$\delta$	mouth opening of crack in multiple-fibre pull-out test
$\bar{\delta}$	effective opening of crack in multiple-fibre pull-out test
$\delta_0$	mouth opening of crack in multiple-fibre pull-out test at onset of fully developed slip
$\bar{\delta}_0$	effective opening of crack in multiple-fibre pull-out test at onset of fully developed slip
$\delta_0^s$	mouth opening of crack in single-fibre pull-out test at onset of fully developed slip
$\bar{\delta}_0^s$	effective opening of crack in single-fibre pull-out test at onset of fully developed slip
$\delta^s$	mouth opening of crack in single-fibre pull-out test
$\bar{\delta}^s$	effective opening of crack in single-fibre pull-out test

$\eta$	toughening ratio	$\tau_v$	maximum attainable shear stress at fibre–matrix interface in single-fibre pull-out test
$\gamma$	constant		
$\phi$	orientation angle of a fibre to a plane perpendicular to the crack	$\tau_v^*$	maximum attainable shear stress at fibre–matrix interface in multiple-fibre pull-out test
$\rho_c$	density of unhydrated cement particles		
$\rho_w$	density of water	$\Theta$	geometry function accounting for the configuration of the discrete phase
$\sigma$	tensile stress		
$\tau$	fibre–matrix interfacial shear stress		
$\tau_g$	shear stress at fibre–matrix interface during pull-out of single fibre	$\varphi$	orientation angle of a fibre to the loading direction
$\tau_g^*$	shear stress at fibre–matrix interface during multiple-fibre pull-out	$\zeta$	ratio $\bar{\delta}_O/\delta_0^S = \bar{v}_{cr}/v_{cr}$