

## Short communication

Fabrication and properties of 3-D C<sub>f</sub>/ZrC–SiC composites by the vapor silicon infiltration processQinggang Li<sup>a,\*</sup>, Shaoming Dong<sup>b</sup>, Zhi Wang<sup>a</sup>, Guopu Shi<sup>a</sup><sup>a</sup>*School of Material Science and Engineering, University of Jinan, Jinan 250022, China*<sup>b</sup>*Structural Ceramics and Composites Engineering Research Center, Shanghai Institute of Ceramics, Chinese Academy of Sciences, Shanghai 200050, China*

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## Abstract

Three-dimensional needled C<sub>f</sub>/ZrC–SiC composites were successfully fabricated by the vapor silicon infiltration process. The bulk density, open porosity, and flexural strength of the composite with PyC/SiC interphases were 2.25 g/cm<sup>3</sup>, 6%, and 301 MPa, respectively. All composites exhibited a non-brittle failure behavior due to propagation and deflection of cracks, as well as fracture and pullout of fibers. The mass loss and linear recession rate of the three-dimensional needled C<sub>f</sub>/ZrC–SiC composite exposed to oxyacetylene torch were 0.013 g/s and 0.004 mm/s, respectively. The formation of ZrSiO<sub>4</sub> melt on the surface of the composite contributed mainly to the excellent high-temperature property of the composite.

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**Keywords:** Vapor silicon infiltration; C<sub>f</sub>/ZrC–SiC composites; High-temperature property

## 1. Introduction

With the demand for application of thermal protection systems, such as rocket nozzles, nosetip, aeronautic jet engines, and leading edges [1,2], carbon fiber-reinforced ultra-high temperature ceramic matrix (C<sub>f</sub>/UHTC) composites have received considerable attention because of their unique physicochemical and mechanical properties. They have been widely used in high-temperature structural elements for aerospace equipments, and can withstand ablation environment with high heat flux and high pressure gas flows because of their high melting points, good thermal-shock resistance and superior ablation/oxidation resistance [3,4].

Continuous fiber-reinforced ceramic matrix composites were fabricated via chemical vapor infiltration (CVI), polymer impregnation and pyrolysis (PIP), reaction sintering (RS), hot pressing, and so on. Compared with CVI and PIP, RS is a more cost effective method for the fabrication of ceramic matrix composites (CMCs). However, RS involves

reactive melt infiltration or liquid silicon infiltration (LSI) [5,6]. LSI has been mostly limited to the fabrication of biomorphic SiC ceramics [7]. Molten silicon infiltration has two drawbacks: reaction choking and flaw generation [8]. However, silicon vapor infiltration has the advantage of infiltrating smaller pores compared with molten silicon infiltration. Furthermore, it can be controlled to avoid reaction choking because of the slow reaction between silicon vapor and carbon [9]. Zhou et al. [9] fabricated carbon fiber-reinforced silicon carbide (C/SiC) composites using the vapor silicon infiltration (VSI) process and obtained dense C/SiC composites.

At present, zirconium carbide (ZrC) is widely used for the fabrication of C<sub>f</sub>/UHTC composites. Our previous study focused on the fabrication of C<sub>f</sub>/ZrC–SiC composites using ZrC precursor and polycarbosilane [10]. Other techniques for the fabrication of C<sub>f</sub>/ZrC–SiC composites have been summarized in our work [11]. Thus far, studies on C<sub>f</sub>/ZrC–SiC composites manufactured through VSI are lacking.

Based on our previous experience, the present work aims to develop a new technique for the fabrication of three-dimensional (3-D) C<sub>f</sub>/ZrC–SiC composites. In this communication, we report our initial studies on the preparation of

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3-D  $C_f/ZrC-SiC$  composites using vapor silicon infiltration. The mechanical and high-temperature properties of the 3-D  $C_f/ZrC-SiC$  composites obtained were studied.

## 2. Experimental procedure

Carbon fibers (T300SC, Toray, Tokyo, Japan) with an average diameter of  $6\ \mu m$  were used. The three-dimensional (3D) fabrics were fabricated by the Nanjing Fiberglass Research and Design Institute (Nanjing, China). The architecture had a fiber distribution of 8:2:1 in the  $x:y:z$  directions, respectively, and an  $\sim 40\%$  fiber volume fraction. According to our previous study [9], a carbon layer (PyC:  $\sim 150\ nm$ ) was firstly deposited on the surface of the carbon fibers to serve as an interfacial layer. The layer was deposited by chemical vapor deposition (CVD) using methane as a precursor. Then, a  $250\ nm$  SiC layer was deposited by CVD, using  $MTS/H_2$  precursor, to protect the fibers from reacting with the silicon vapors. Commercially available ZrC particles (Kaier Nanometer Technology Development Co. Ltd., Hefei, China) were balled with a phenolic resin (Shanghai Qinan Adhesive Material Factory, Shanghai, China) at a certain ratio, using ethyl alcohol as solvent to form homogenously dispersed slurry. Then the fabrics were impregnated by the aforementioned slurry, followed by pyrolysis at  $900\ ^\circ C$  to form a porous body ( $C_f/ZrC-C$  composites). The slurry infiltration process was performed in a vacuum, and then a pressure of  $2\ MPa$  was applied by nitrogen gas to facilitate the infiltration process. The pyrolyzed body contained a carbon matrix, ZrC, and a distribution of microspores. Such composites contained large number of pores and cracks. In order to fabricate  $C_f/ZrC-SiC$  composites by the VSI process, a silicon powder (purity:  $99.9\%$ , grain size:  $75\ \mu m$ , Sinopharm Chemical Reagent Co. Ltd., Shanghai, China) was put inside a graphite crucible, then the  $C_f/ZrC-C$  composites. Finally, the graphite crucible with silicon powder and  $C_f/ZrC-C$  composite were sealed and heated up to  $1700\ ^\circ C$  for  $1.5\ h$  in vacuum (residual pressure  $\sim 1\ Pa$ ). The gaseous Si penetrated into the porous channel and concurrently facilitated the reaction to form SiC. As a result, the 3-D  $C_f/ZrC-SiC$  composite was fabricated.

The densities of the samples were measured using Archimedes' method in water after evacuating the air from porosity. The samples were cut into  $4\ mm \times 5\ mm \times 60\ mm$  specimens and polished for three-point bending test. The flexural strength was measured by the three-point bending test in an Instron-5566 (Instron Corp., Canton, MA) universal testing machine, with a crosshead speed of  $0.5\ mm/min$  and a span of  $48\ mm$ . Young's modulus was calculated from the data recorded during three-point bending testing [12].

The phase compositions of the composites were characterized by X-ray diffraction (XRD, Mode: RAX-10, Rigaku, Japan) with Cu  $K_\alpha$  radiation. The microstructures of the composites were studied by Electron Probe Micro-analyzer (EPMA, JXA-800, Jeol, Tokyo, Japan).

The oxidation resistance of the composites was evaluated by static oxidation at  $1700\ ^\circ C$  for  $15\ min$  in a muffle furnace. The anti-ablation property test was conducted in a flowing oxyacetylene torch environment. During the test, a specimen with a size of  $80\ mm \times 80\ mm \times 10\ mm$  was vertically exposed to the flame for  $600\ s$  when the surface temperature of the composite reached  $1800\ ^\circ C$ . The distance between the nozzle tip and the surface of the specimen was  $10\ mm$  and the inner diameter of the nozzle tip was  $2.0\ mm$ .

## 3. Results and discussions

The density and open porosity of the 3-D  $C_f/ZrC-SiC$  composite with PyC/SiC interphase before and after VSI are shown in Table 1. It can be seen that there are obvious differences observed among the composites before and after VSI. After VSI the bulk density of the composite increased from  $1.77\ g/cm^3$  to  $2.25\ g/cm^3$ , whereas its open porosity decreased from  $24\%$  to  $6\%$ . By comparison, it is concluded that open porosities provide channels for the VSI process. Due to the VSI process, the density increased and open porosity decreased. During the VSI process, silicon vapor entered into the channels and the reaction generated SiC filled open porosities. On the whole, VSI can be considered as a suitable method for fabrication of dense composites, which is the same as our previous work [13]. In order to guarantee the VSI process smoothly and the amount of SiC matrix formed by the reaction between

Table 1  
Density and open porosity of 3-D  $C_f/ZrC-SiC$  composite with PyC/SiC interphase before and after VSI.

| Sample        | Density ( $g\ cm^{-3}$ ) | Open porosity (%) |
|---------------|--------------------------|-------------------|
| $C_f/ZrC-C$   | 1.77                     | 24                |
| $C_f/ZrC-SiC$ | 2.25                     | 6                 |

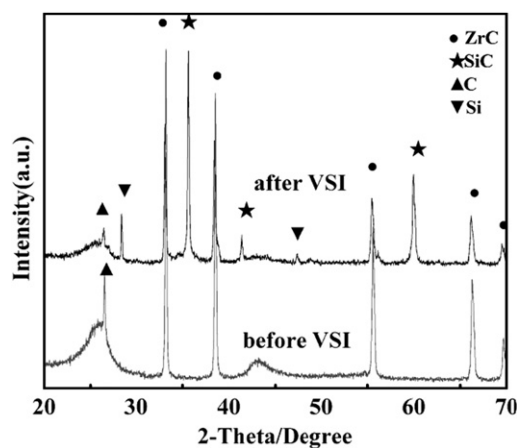


Fig. 1. XRD patterns of 3-D  $C_f/ZrC-SiC$  composites with PyC/SiC interphase before and after VSI.

silicon and carbon, the open porosity before VSI is controlled about 20%.

The XRD patterns of the 3-D  $C_f/ZrC-SiC$  composites with PyC/SiC interphase before and after VSI are plotted in Fig. 1. A significant difference was found in the XRD patterns of the 3-D  $C_f/ZrC-SiC$  composites before and after VSI. The diffraction peaks mainly belong to C and ZrC before VSI, whereas diffraction peaks correspond to ZrC, SiC, Si, and C. The reaction between silicon vapor and carbon is dependent on the temperature of infiltration and the concentration of silicon vapor [14]. At 1700 °C, the silicon vapor became abundant. Some reacted with carbon forming SiC matrix, and the rest remained in the matrix. The forming SiC also made the density and open porosity in Table 1 reasonable.

At 1700 °C, a larger amount of vapor silicon reacted with the carbon and resulted in denser composites with better physical and mechanical behavior when the interphases were deposited, as shown in Table 2. The samples with interphases had a higher density, showing that relatively dense matrix can be achieved by VSI. Only a

small difference was noted in their open porosity. It is also concluded that VSI is effective method to density the composites, and that the composites have good mechanical properties when the interphases are deposited.

The micrographs of polished cross-sections and EDS analysis for composites without interphase before and after VSI are shown in Fig. 2. It is obvious that some pores are observed not only in the intrabundle areas but also in the interbundle areas for the composite before VSI. However, for the composite after VSI, the interbundle area shows densified microstructure even though some closed micropores are observed in the intrabundle areas as shown in Fig. 2(b). As we know, it was found that fibers within a yarn were adhering to each other, resulting in the formation of closed micropores after impregnation. Thus, during the VSI process, the silicon vapor is difficult to infiltrate into the closed micropores in the intrabundle areas. Fig. 2 also shows the EDS analyses of a composite fabricated by the VSI process. Based on the EDS analyses, it shows the presence of Zr and C before VSI and Zr, C and Si after VSI. It is indicated that SiC matrix is formed and that the

Table 2  
Properties of 3-D  $C_f/ZrC-SiC$  composites with different interphases.

| Interphase | Fiber fraction | ZrC fraction | Density (g/cm <sup>3</sup> ) | Open porosity | Bending stress (MPa) | Elastic modulus (GPa) |
|------------|----------------|--------------|------------------------------|---------------|----------------------|-----------------------|
| None       | 28 vol%        | 26 vol%      | 2.01                         | 10%           | 121 ± 13             | 17 ± 2.0              |
| PyC        | 27 vol%        | 22 vol%      | 2.20                         | 8%            | 221 ± 27             | 23 ± 3.0              |
| PyC/SiC    | 25 vol%        | 24 vol%      | 2.25                         | 6%            | 301 ± 15             | 25 ± 4.0              |

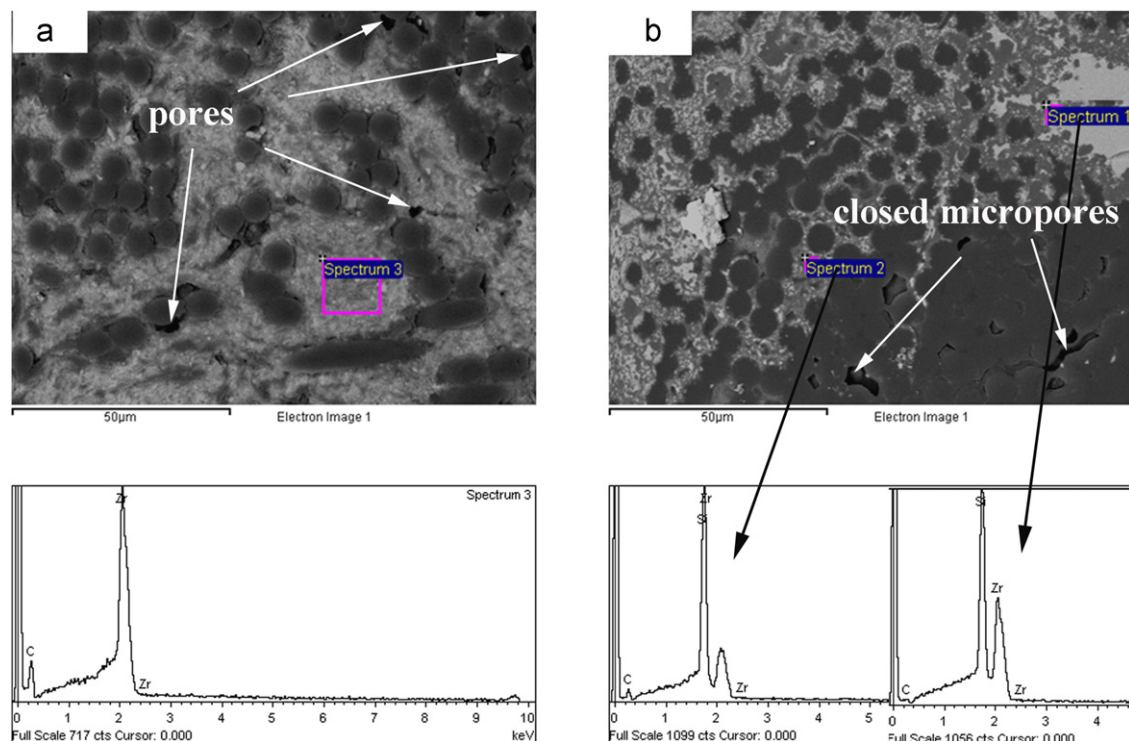


Fig. 2. SEM micrographs on the polished cross-section of 3-D  $C_f/ZrC-SiC$  composite without interphase: (a) before VSI and (b) after VSI.

residual Si existed after the VSI process according to the EDS analyses (Fig. 2) and the XRD patterns of the composites (Fig. 1). The residual Si in composites is attributed to the silicon vapor condensation during the cooling of the VSI process. When there are no interphases between the carbon fiber and matrix, the silicon vapor reacts not only with the carbon matrix, but also with carbon fibers, as shown in Fig. 2(b).

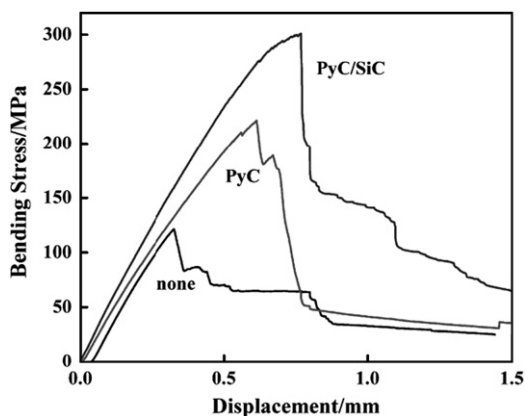


Fig. 3. Bending stress/displacement curves of 3-D  $C_f/ZrC-SiC$  composites with different interphases.

The bending stress–displacement curves of all the composites with different interphases are shown in Fig. 3. Based on these curves, all samples showed a typical non-brittle fracture behavior regardless of the existence of interphases. With the deposition of interphases, mechanical properties including elastic modulus and bending stress increased. This result is consistent with the data listed in Table 2. As mentioned before, the reaction between carbon fiber and silicon could decrease the fiber strength and increase the bonding strength between fiber and matrix.

The morphologies of fracture surfaces are shown in Fig. 4. It can be seen that all the composites showed a typical non-brittle fracture behavior, and fiber pullouts accompanied the fracture process. When no interphase was deposited, short pulled-out fibers with rather coarse surfaces can be observed. By contrast, with PyC/SiC or PyC interphases, longer pulled-out fibers with smoother surfaces can be observed. This phenomenon can be explained by the reaction between the fibers and the silicon vapor. When no interphase was deposited, silicon vapor reacted with the fibers, which damaged the fiber surfaces. As a result, some matrices evenly clung to the pulled-out fibers. However, the fibers can be protected from reacting with silicon vapor at such high temperature. As shown in Fig. 4(b) and (c), the fiber surfaces remained intact, and crack deflection parallel to the fiber axis can also be detected.

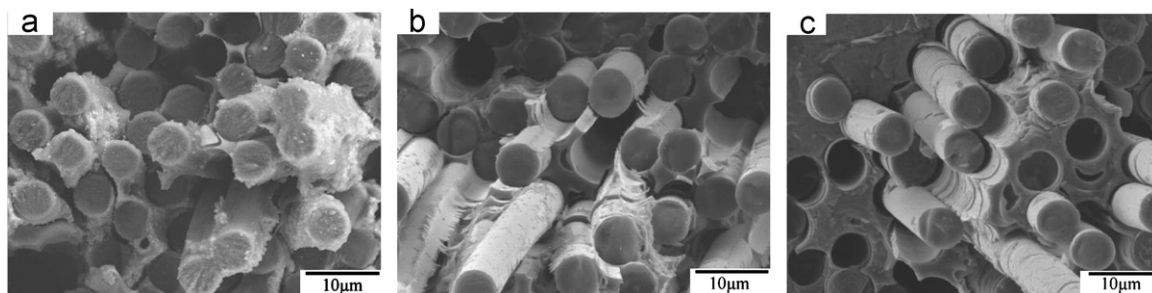


Fig. 4. SEM micrographs on the fracture surfaces of 3-D  $C_f/ZrC-SiC$  composites: (a) none, (b) PyC and (c) PyC/SiC.

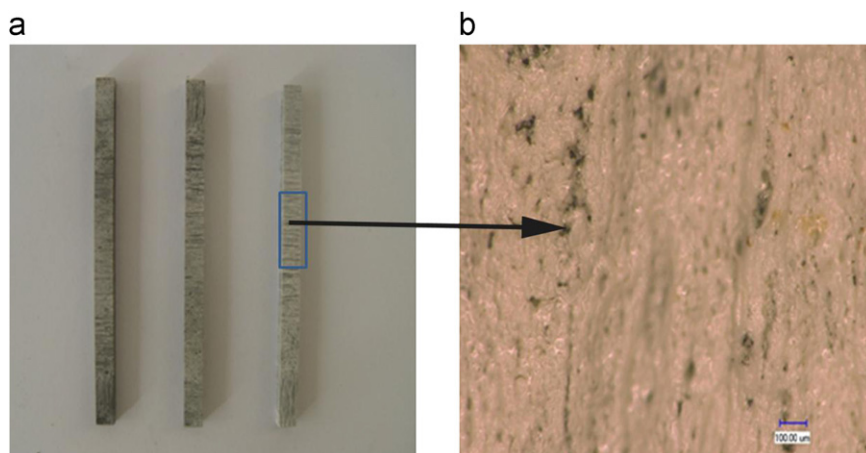


Fig. 5. Optical images of 3-D  $C_f/ZrC-SiC$  composites after oxidation: (a) after oxidation and (b) larger magnification.



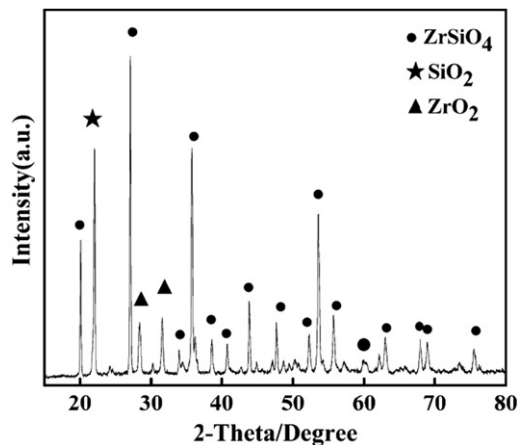


Fig. 6. XRD pattern of 3-D  $C_f/ZrC-SiC$  composite surface after oxidation.

The high temperature property of the  $C_f/ZrC-SiC$  composite is important. Hence, the oxidation resistance of the composites was evaluated through static oxidation at 1700 °C for 15 min in a muffle furnace. The optical images and XRD patterns of the 3-D  $C_f/ZrC-SiC$  composite after oxidation are shown in Figs. 5 and 6, respectively. As shown in Fig. 5(a), the shape of the 3-D  $C_f/ZrC-SiC$  composite after oxidation remained intact and the surface was totally covered with oxide. As shown in Figs. 5(b) and 6, a glass layer of  $ZrSiO_4$  was formed when used in oxidative atmosphere. The phases include  $ZrSiO_4$ ,  $SiO_2$ , and  $ZrO_2$ . The glass layer of  $ZrSiO_4$  would prevent the composite from further oxidation. Meanwhile, the mass loss and linear recession rate of the  $C_f/ZrC-SiC$  composites exposed to oxyacetylene torch were 0.013 g/s and 0.004 mm/s, respectively. Compared with our previous research results [15,16], the mass loss and linear recession rate of the  $C_f/ZrC-SiC$  composites are almost the same. Therefore, 3-D  $C_f/ZrC-SiC$  composites have excellent high-temperature properties.

#### 4. Conclusions

Three-dimensional  $C_f/ZrC-SiC$  composites were fabricated via the VSI process. The bulk density, open porosity, and flexural strength of the composite with  $PyC/SiC$  interphases were 2.25 g/cm<sup>3</sup>, 6%, and 301 MPa, respectively. All composites exhibited a non-brittle failure behavior. Based on the oxidation resistance and ablation test through oxyacetylene torch, the results indicate that 3-D  $C_f/ZrC-SiC$  composites have excellent high-temperature properties.

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